

NAVAL POSTGRADUATE SCHOOL Monterey, California



THESIS



A DIGITAL FILTER REPRESENTATION OF THE ASQ-81 MAGNETOMETER

by

Michael Charles Huete

September 1983

Thesis Advisor:

Andrew R. Ochadlick

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A Digital Filter Representation of the ASQ-81
Magnetometer

bу

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Lieutenant, United States Navy
B. S. E. E., Tulane University, 1976

Submitted in partial fulfillment of the requirements for the degree of

MASTER OF SCIENCE IN SYSTEMS TECHNOLOGY (Antisubmarine Warfare)

from the

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September 1983

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ABSTRACT

digital filter representation of the ASQ-81 magnetometer is derived from the s-plane transfer functions of the system through the use of a bilinear transformation. A FORTRAN computer program is written which applies representation to time-sampled total magnetic field data order to obtain a time series representation of ASQ-81 filtered total field. A series of simulations and a field experiment are conducted which verify the program output. Applications of this program include usage in conjunction with geomagnetic field data in order to produce a new data set representative of geomagnetic noise observed by Navy MAD (Magnetic Anomaly Detection) aircraft with the potential to investigate techniques of reducing geomagnetic noise in MAD aircraft.

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I. INTRODUCTION

The detection and location of submarines (and other magnetic bodies) through the discrimination of changes or anomalies in the Earth's magnetic field is called Magnetic Anomaly Detection or MAD. In this technique, a magnetometer measures the magnitude of the Earth's magnetic field and provides an indication of that magnitude, or, more usually, an indication of changes in the magnitude of the Earth's field. These changes, or anomalies, can indicate the presence of magnetized bodies which may or may not be a submarine.

The magnetometer currently in use in the United States Navy for use in this MAD process is the AN/ASQ-81 metastable helium vapor total field magnetometer.

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Research is currently being conducted at the Naval Postgraduate School in Monterey, California, in various aspects of the applications of magnetometers, including Magnetic Anomaly Detection (MAD). Within the context of this research, magnetic field measurements are made through the use of sets of wire wound coils vice any specific magnetometer or magnetic detecting system. The data collected through the use of these coils is evaluated and

processed in a variety of methods for different project goals.

thesis project is designed to produce an acceptable alternative to the physical presence of an experimental AN/ASQ-81 magnetometer at the postgraduate school the determination, in conjunction with other progress, of the output of the AN/ASQ-81 in magnetometer from the data collected from the school's measurement coils. It is hoped that this will assist future projects as, for example, in allowing determination of environmental noise of such characteristics to affect the AN/ASQ-81 magnetometer operationally with the eventual goal of providing an environmental noise index a system of removing such noise from the magnetometerdetection system.

II. GEOMAGNETICS REVIEW

A. EARTH'S MAGNETIC FIELD

1. Constituents of the Geomagnetic Field

The most common method of specifying the consituting parts of the geomagnetic field is to divide the field in terms of distance from the center of the Earth. This method results in three classifications: <u>internal</u>, <u>crustal</u>, and <u>external</u>. [Ref. 1]

The internal field originates in the core region and is the most stable field, containing only extremely low frequency temporal variations. The crustal, or anomalous, field arises from modifications made on the internal field by materials and structures in the Earth's crust. These variations are not constant with regard to spatial locations, and comprise part of what is known as geological variations. The external field is the most dynamic and arises from many sources, including the interaction between the solar wind and the Earth's magnetic field.

In addition to this method of defining the Earth's magnetic field is the method of time variations. This method consists of considering that part of the field which varies with periodicities greater than about one year as the

steady field and everything else as the variation field.
[Ref. 2]

The steady field consists of the <u>internal field</u>, also referred to as the <u>main field</u>. Slow variations of the main field with periods of years or longer are referred to as <u>secular variations</u>.

There are various elements that contribute to the geomagnetic field, some of which are external to the Earth's surface. External contributions make up only a small part of the steady field, but play an important role in the variation field. These external sources include current systems in the Earth's upper atmosphere affected by solar electromagnetic radiation and gravitation, solar corpuscular radiation and the interaction of solar plasma with the main field, and the effect of the solar inteplanetary field.

[Ref. 3]

The geomagnetic field changes with time. As previously mentioned, very slow variations in the main field with periods of on the order of years to thousands of years are referred to as secular variations. Secular variations are caused by a variation in the strength or orientation of the Earth's center dipole.

Other time variations of the field can be categorized into quiet variation fields and disturbed variation fields.

Disturbed variation fields include geomagnetic micropulsations, which are of particular interest to

operational forces as these can mask target signatures and are therefore a source of noise to MAD sensors.

Quiet variation fields are those which are not due to disturbances in the interplanetary environment and which vary slowly and regularly. [Ref. 3]

Disturbed variation fields are geomagnetic field variations that appear to be the result of interplanetary environmental changes and do not posess These variations include ionospheric periodicity. disturbances, the aurora, geomagnetic storms, geomagnetic micropulsations.

2. Elements of the Magnetic Field Vector

The geomagnetic field vector is characterized at any point by its direction and magnitude. This is commonly accomplished through a system of coordinates as shown in Figure 2.1. The field is measured in terms of local coordinates with respect to true North. [Ref. 3]

The various coordinates are referred to as <u>magnetic</u> <u>elements</u> and are defined as follows:

- B: Total field intensity (the symbol \underline{F} is sometimes also used, as in this figure.)
- H: Horizontal component
- X: Northward, or NorthSouth component
- Y: Eastward, or EastWest component
- Z: Downward, or Vertical component

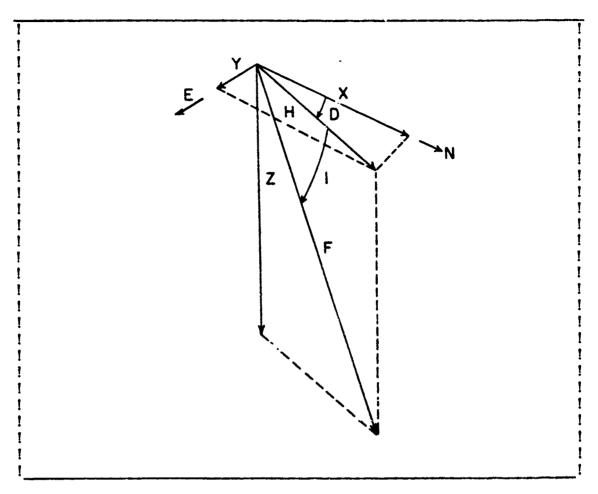


Figure 2.1 Magnetic Field Elements [Ref. 4].

- I Inclination or Dip Angle. This is the angle between \underline{H} and \underline{B} (or \underline{F}) and is measured positive downward.

III. THE AN/ASQ-81 MAGNETOMETER

A. DESCRIPTION OF SYSTEM OPERATION

Magnetic Anomaly Detecting set currently in use the U S Navy is the AN/ASQ-81 magnetometer. This used to locate and classify submerged submarines by sensing the Earth's magnetic field (anomalies) disturbances in caused by the presence of the magnetic mass of submarine. The disturbance of the Earth's field is detected by the magnetometer, processed through filtering circuits, The output signal of the magnetometer and amplified. displayed on a chart recorder for interpretation operator.

The magnetic detecting set is a metastable helium magnetometer. The operation of the magnetometer is based on the light absorbtion properties of helium gas subjected to certain light stimulus (optical pumping), radio frequency Earth's excitation, and the magnetic field. The magnetometer consists of a helium lamp, lens and polarizer generate a beam of polarized light radiation. focused and polarized light beam is directed through a helium absorbtion cell to an infrared (IR) detector. the helium gas in the absorbtion cell is maintained in a metastable state by application of VHF excitation.

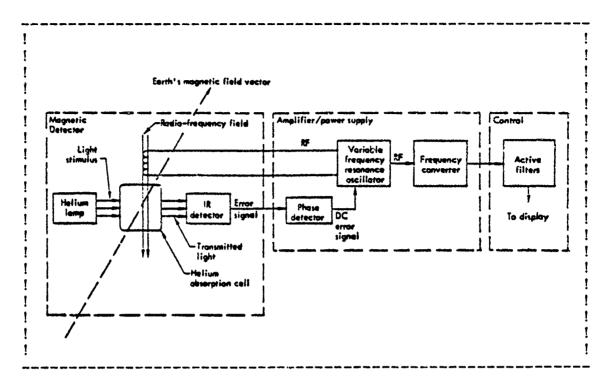


Figure 3.1: Metastable Helium Magnetometer [Ref. 5].

The Earth's magnetic field imposes a magnetic force the excited helium vapor atoms to force the atoms into three energy sublevels. This is called the effect. The rate or frequency of atomic precession caused by this effect is called the Larmor frequency. lamp is used to optically pump the atoms in the absorbtion cell, with the result that the polarized light passing through the absorbtion cell will polarize (magnetize) the helium atoms in the absorbtion cells selectively pumping the Zeeman levels of the energy of helium atoms in the cell. The magnetization direction determined bу the polarization of the photons from the helium lamp.

is then introduced to the absorbtion cell RF the form of an additional magnetic field imposed through the use of coils oriented perpendicular to the precessed polarized helium atoms in the absorbtion cell and energized by a variable frequency RF oscillator. The RF oscillator is frequency, which results the Larmor in to depolarization of the atoms. The atoms attempt to equally repopulate the Zeeman energy levels. However, the helium lamp is still beaming polarized light energy into the absorbtion cell, causing the atoms to absorb light energy and rise to an excited energy levbel. This absorbtion of light energy is detected through the use of an infrared RF oscillator frequency producing detector. The absorbtion is called the resonant frequency, determined through the use of a servo loop from the infrared detector to the RF variable frequency oscillator.

Therefore, any change in the Earth's magnetic intensity will result in a change in the Larmor frequency of the helium atoms in the helium absorbtion cell. This new Larmor frequency willbe detected bу ASQ-81 Since the gyromagnetic ratio of helium magnetometer. 28.024 HZ per gamma, this detection of the resonant frequency provides a measurement of the Earth's magnetic field intensity at any given time. A change in the Earth's magnetic field intensity could signal the presence of a submerged submarine.

The output resonant frequency developed by the magnetometer is converted to a proportional output voltage which is filtered through the Magnetic Anomaly Detection (MAD) bandpass filters for environmental noise reduction and utilized to drive a chart recorder for observation by an operator. [Ref. 6]

B. TRANSFER FUNCTIONS

Transfer functions for the AN/ASQ-81 filters obtained from the manufacturer of the AN/ASQ-81 detecting set, Texas Instruments of Dallas, Texas. These transfer functions are listed in Appendix A and are in the form of H(s), that is, the frequency domain, or S domain, where S = jw. The s-domain representation for transfer functions routinely utilized is to express output system characteristics for given system inputs. As the S domain representation is not utilzed further in this discussion, it will not be further explained.

As the output signal of the ASQ-81 magnetometer is filtered through a fixed high-pass system, then through a selectable low pass system and a selectable high pass system (as shown in Figure 3.2 below), the transfer funtions are listed in this order.

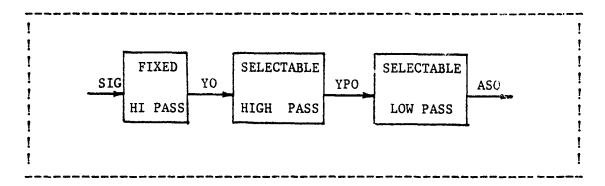


Figure 3.2 : Signal Flow Diagram for ASQ-81

Further discussion will be made of the selectable filters later.

IV. DIGITAL FILTERING MODELLING OF SYSTEMS

A. SEQUENCE REPRESENTATION OF TIME FUNCTIONS

1. Signal Representation

A <u>signal</u> can be defined as a function that conveys information, generally about the state or behavior of a physical system. Although signals can be represented in many ways, the information conveyed by the signal is contained in a pattern of variations of some form. Signals are represented mathematically as functions of one or more independent variables, one of the most common of which is time.

The independent variable of the mathematical representation of a signal may be continuous or discrete. Continuous time signals are signals that are defined over continually values of time and are therefore represented by continuous-variabled functions. Discrete time signals are defined at discrete time intervals and are therefore represented by functions whose independent variable(s) take on discrete values only. Discrete-time signals are represented as sequences of numbers. [Ref. 7]

In addition to the fact that the independent variables can be either continuous or discrete, the signal amplitude can be either continuous or discrete. Digital

signals are those for which both time and amplitude are discrete. Analog signals are those for which both time and amplitude are continuous.

Digital signal processing deals with transformations of signals that are discrete in both time and amplitude, usually represented by sequences of numbers. The <u>nth</u> number in the sequence \underline{x} being processed is usually represented as $\underline{x}(\underline{n})$, and is formally written as:

$$x=[x(n)], -\infty < n < +\infty$$

In general, an arbitrary sequence can be expressed as

$$x(n) = \sum_{k=-\infty}^{\infty} x(k) d(n-k)$$

where d(n-k) is the <u>unit sample</u> at time k. In other words, an arbitrary sequence may be expressed as a sum of scaled, shifted unit samples, where the scaling factor is equal to the amplitude of the sequence at that time.

2. Linear Shift-Invariant Systems

A <u>system</u> is defined mathematically as a unique transformation or operator that maps an input sequence [x(n)] into an output sequence [y(n)]. This is denoted as:

$$y(n) = T[x(n)]$$

and is often depicted as in Figure 4.1.

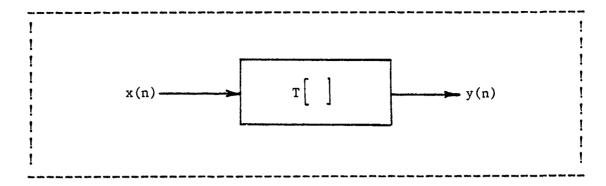


Figure 4.1: Representation of Transformation of an Input Sequence to an Output Sequence. [Ref. 7]

Classes of discrete time systems are defined by placing constraints on the transformation $T[\]$.

The class of <u>linear systems</u> is defined by the principle of superposition. If y (n) and y (n) are the 1 2 responses when x (n) and x (n) are the inputs, then a system 1 2 is linear if

$$T[ax(n) + bx(n)] = aT[x(n)] + bT[x(n)]$$

$$1 2 1 2$$

$$= ay(n) + by(n)$$

$$1 2$$

for any arbitrary constants a and b. This, together with the concept of representing a sequence by a sum of delayed and scaled unit-sample sequences, suggests that a linear system can be characterized by its unit-sample response. Specifically, let h(n) be the response of the system to k

$$y(n) = T[\sum_{k=-\infty}^{\infty} x(k) d(n-k)]$$
 or

$$y(n) = \sum_{k=-\infty}^{\infty} x(k) T[d(n-k)] = \sum_{k=-\infty}^{\infty} x(k) h(n)$$

Thus the system response can be expressed in terms of the response of the system to d(n-k).

The class of <u>shift invariant systems</u> is characterized by the property that if y(n) is the response to x(n), then y(n-k) is the response to x(n-k), where k is a positive or negative integer. When the index n is associated with time, shift-invariance corresponds to <u>time-invariance</u>. The property of shift invariance implies that if h(n) is the response to d(n), then the response to d(n-k) is simply h(n-k). Therefore

$$y(n) = \sum_{k=-\infty}^{\infty} x(k) h(n-k)$$

and any linear shift-invariant system is completely characterized by its un':-sample response h(n).

A subclass of linear shift-invariant systems consists of those systems for which the input $\mathbf{x}(n)$ and the output $\mathbf{y}(n)$ satisfy an Nth-order linear constant-coefficient difference equation of the form

$$\sum_{k=0}^{N} a_k y(n-k) = \sum_{r=0}^{M} b_r x(n-r)$$

If the assumption of causality is made about the system, a linear difference equation provides an explicit relationship between the input to the system and the output of the system. This can be seen by rewriting the previous equation as

$$y(n) = \sum_{k=1}^{N} c_k y(n-k) + \sum_{r=0}^{M} d_r x(n-r)$$

where c = -a/a and c = b/a.

Thus the nth value of the output can be computed from the nth value of the input and the N and M past values of the output and input, respectively. The difference equation not only represents the system for theoretical purposes, but it may also serve as a computational realization of the system. The z-Transform makes use of this property to realize systems.

B. THE z-TRANSFORM

1. Description of the z-Transform

The <u>z-transform</u> plays an important role in the analysis and representation of discrete-time linear shift-invariant systems. The z-transform, X(z), of a sequence x(n) is defined as

$$X(z) = \sum_{n=-\infty}^{\infty} x(n)z^{-n}$$

where z is a complex variable. This representation of the z-transform is referred to as the <u>two-sided z transform</u>. The <u>one sided z-transform</u> consists of the same summation for terms of n greater than or equal to zero. For the case that x(n)=0 for n<0, the one sided and the sided z transforms are equivalent.

By expressing the complex variable z in polar form as $z = re^{-x}$, the z-transform can be interpreted as the Fourier transform of x(n) multiplied by an exponential sequence. For r = 1, that is, for |z| = 1, the z-transform is equal to the Fourier transform of the sequence.

2. The Bilinear Transformation

The transfer functions of analog systems are most often expressed in terms of s = jw (see section III B.). This corresponds to the analog frequency response of This analog frequency response can be "mapped", system. that is, tranformed to the z-plane from the s-plane through the bilinear transformation. use of The effect of utilizing the bilinear transformation is to convert a system transfer function in terms of the variable S into the system transfer function in terms of the variable The transformation itself is:

and

$$z = \frac{(2/T) + s}{(2/T) - s}$$

where T is the sampling period, that is, the time between data samples.

Thus a transform can be made from one plane to the other. In this manner, the transfer function, H(z), of a system may be obtained.

The bilinear tranformation equations may be shown to hold in general, and the use of this transformation may be shown to yield stable digital filters from stable analog filters [Ref. 7]. The bilinear transformation maps the imaginary jw axis in the s-plane onto a unit circle (of the region of convergence) in the z-plane, with the left half s-plane mapped onto the region inside the circle and the right hand (region of instability) s-plane mapped onto the region outside this circle [Ref. 8]. A complete discussion of the z-transform is available in several texts, some of which are listed in the Bibliography.

C. THE DIGITAL COMPUTATIONAL ALGORITHM

In implementing a digital filter on a digital computer such as the IBM 3033, the input-output relationship of the signals through the system being synthesized must be converted to a computational algorithm. The algorithm is specified in terms of a set of basic computations of elements. For the implementation of discrete-time systems

described by linear constant coefficient difference equations, such as the AN/ASQ-81, it is convenient to choose as these elements the basic operations of addition, delay, and multiplication by a constant. The computational algorithm for implementing the filter is then defined by a structure or network consisting of an interconnection of these basic operations. For a system transfer function of the form

$$H(z) = \frac{\sum_{k=0}^{M} b_{z}^{-k}}{\sum_{k=1}^{N} a_{z}^{-k}}$$

$$1 - \sum_{k=1}^{N} a_{z}^{-k}$$

the difference equation relating input and output is easily written down directly from the system function and is given by $\,$ N $\,$ M

$$y(n) = \sum_{k=1}^{N} a_k y(n-k) + \sum_{k=0}^{M} b_k x(n-k) \quad [Ref. 7]$$

This difference equation can be interpreted directly as a computational algorithm in which the delayed values of the input are multiplied by the coefficients b, the delayed values of the output are multiplied by the coefficients a, and the resulting products are added. It is now easy k to see the process to be followed in obtaining the computational algorithm for the AN/ASQ-81 magnetometer

transfer function. The z-transform of the system transfer function is obtained through the use of the bilinear transformation, and is then converted into a difference equation relating input and output signals, thence to a FORTRAN computer program. A table of z-transforms of system functions is included in Appendix B.

In the FORTRAN computer program realization of the total system computational algorithm, each filter block is transformed into a separate difference equation and algorithm. This was done to enable a "building block" type approach to the program, and to minimize computational and roundoff errors.

D. THE CASCADE FORM OF THE COMPUTATIONAL ALGORITHM

Even though the direct form realization of the digital filter design may be perfectly satisfactory in a theoretical sense, it may be less than desirable in the context of realization through the use of a general purpose computer of fixed register length. The parameters of a digital filter are usually obtained with a high degree of accuracy, which results in a faithful realization of the desired system. When these parameters are quantized, as in a finite memory register within a computer, the frequency response of the resulting digital filter may differ appreciably from the original design. In fact, the quantized filter may fail to

meet design specifications although the unquantized filter does. [Ref. 7]

The sensitivity of the filter response to errors in the filter parameters is dependent upon the structure of the filter realization. Therefore, in the event of an unacceptable change in the frequency response of the filter due to quantization errors, it is often possible to minimize the effect of these errors through an alternate filter realization structure. An alternate structure to the previoulsy discussed direct form realization is the <u>cascade form</u> realization.

The direct form network structures were obtained directly from the system function H(z) written in the form of a ratio of sums. If this ratio is factored into a product of polynomials of the form

this product represents a general distribution of poles and zeros and suggests a set of structures consisting of a cascade of first and second-order subsystems. There is considerable freedom in the design of the subsystems, but it is best to realize the systems using a minimum of storage.

The expression of H(z) in this form indicates the presence of poles and zeros in pairs. If poles and zeros

are not present in pairs, one of the coefficients B or a 2k 2k will be zero as appropriate. An implementation of such a cascade structure with the use of minimum memory can be obtained through a direct form II realization of each second order subsystem using techniques similar to the direct form implementation utilized previously. A cascade realization of a sixth-order system, such as the ASQ-81 system, using a direct form II realization of each second order subsystem would appear as in Figure 4.2 below.

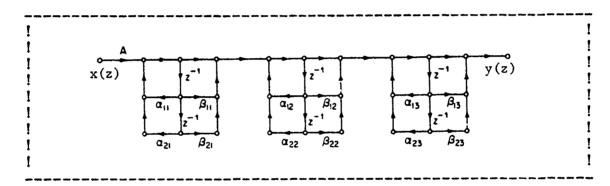


Figure 4.2: Cascade Structure With a Direct Form II Realization of Each Second Order Subsystem. [Ref. 7]

There is, theoretically, considerable flexibility in the manner in which the poles and zeros are paired together and in the order in which the resulting second-order subsystems are cascaded. However, although all such pairings and orderings are equivalent for infinite-precision arithmetic, they may differ considerably in practice owing to finite word length effects of roundoff and truncation.

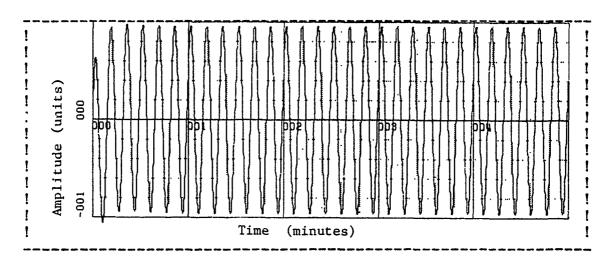


Figure 5.1: Output of First Stage Filter of Digital Filter Computer Program With Sinusoidal Input in Simulation.

Unfortunately, the second stage output of the filter showed an instability within the program design, indicated by the output of the filter being a sinusoid of increasing magnitude, as indicated in Figure 5.2 below.

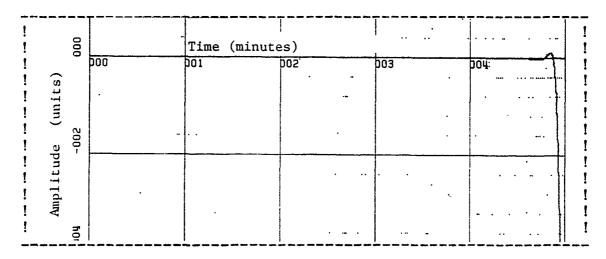


Figure 5.2: Output of Second Stage Filter Design With Input of a Sinusoid.

The stability of the third stage of the filter design was investigated by inputting the sinusoid directly to the third filter, and found to be stable. A check of the derivation of the equations, coefficients, and programming steps of the second (unstable) filter of the design failed to indicate the cause of the instability.

Computation of the poles of the z transfer function, H(z), of the second stage of the filter confirmed the instability of the design. The poles were computed to be: $0.92 \pm 0.1218i$, $1.07 \pm 0.1340i$, 0.8611, and 1.1564. Of these six poles, three lie outside the region of convergence for the z-plane, that is, within the unit circle discussed previously in Chapter IV.

The second stage of the filter was therefore redesigned using the cascade form of the direct form realization (direct form II), and tested in simulation. A copy of the software used in the simulation is enclosed in Appendix F.

The output of all three filter stages of the program were stable, as indicated in Figures 5.3 through 5.7 below. The amplitude decrease and phase shift expected were observed. The "damped overshoot" of the second stage output is due to the fact that, for values of the input function prior to time zero in the simulation, utilized in the input output signal difference equations for the filter, the input signal was set at 0. This resulted in an instantaneous

change of the input signal from 0 to the finite value introduced in the simulation at time 0+. The "overshoot" of the filter is the filter's attempt to "match" this instantaneous jump in magnitude of the input signal. When the input signal to the filter in the simulation is zero at time zero, this overshoot effect does not occur.

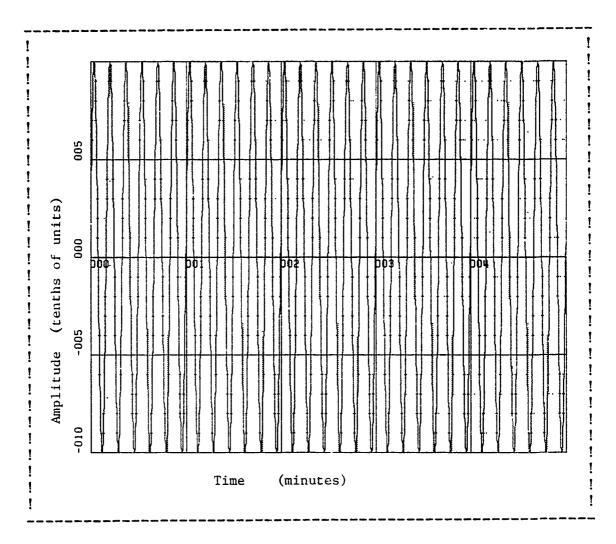


Figure 5.3: Input Signal to Digital Filter Program. A Sinusoid of Frequency 0.1 HZ and Amplitude \pm 1.

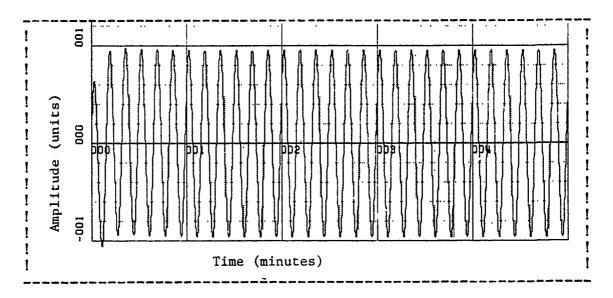


Figure 5.4: Output of First Stage of Digital Filter Program.

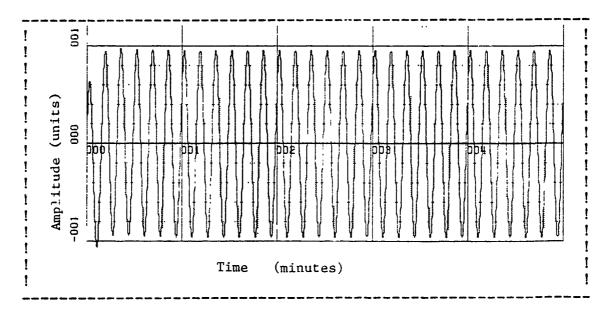


Figure 5.5: Output of Second Stage of Digital Filter Program.

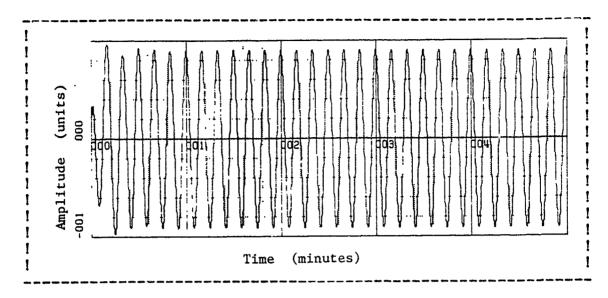


Figure 5.6: Output of Third Stage of Digital Filter Program.

This simulation was run with inputs of sinusoids of various frequencies in order to check the stability of the filter design at frequencies throughout the operating range of the AN/ASQ-81 magnetometer. In all cases, the design was stable, and the expected amplitude changes and phase shifts occured.

2. Noiselike Inputs

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The simulation was also run with inputs consisting of a sinusoid of a frequency which should be passed through the AN/ASQ-81 added to sinusoids of frequencies which should have been filtered by the magnetometer and random noise. The filter performed as expected, with the sinusoid of a passable frequency passed by the filter, and spurious noise and sinusoids attenuated severely. The results of a simulation consisting of a sinusoid of passable frequency, a

filterable sinusoid, and uniformly distributed random noise, all of amplitude ± 1 , are presented in Figures 5.7 through 5.10 below.

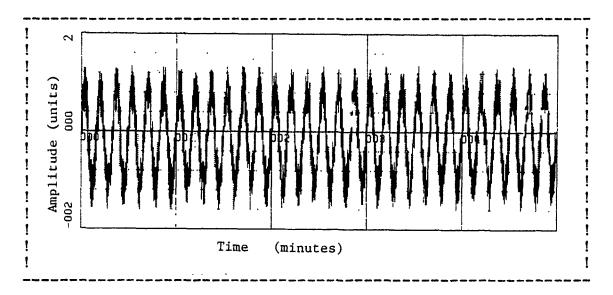


Figure 5.7: Input to Filter - 0.1 HZ Sinusoid, 10 HZ Sinusoid, Uniformly Distributed Random Noise of Amplitude ± 1 .

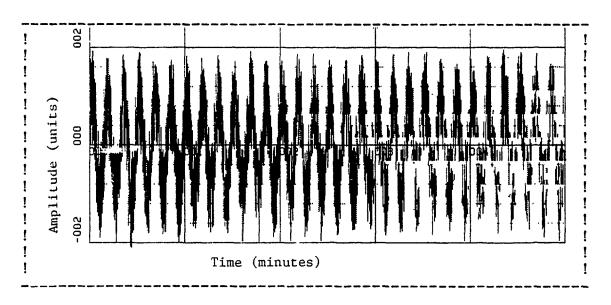


Figure 5.8: Output of First Filter Stage.

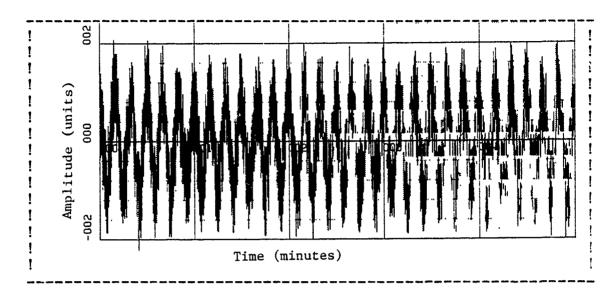


Figure 5.9: Output of Second Filter Stage.

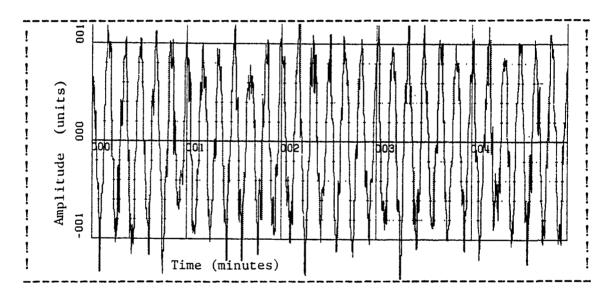


Figure 5.10: Final Filter Stage Output.

As can be seen, the digital filter program succeeds in filtering out random noise and signals of frequency components above the band pass of the magnetometer.

In order to ensure that the digital filter representation of the magnetometer has the same amplitude

AN/ASQ-81 versus frequency characteristics of the magnetometer, a simulation program was written which inputs sinusoids of varying frequencies and computes the Root Mean Square (RMS) value of the filter output and the then computes the decibel (dB) attenuation of input, filter at that frequency. A copy of this program included in Appendix G. A plot was made of the dB attenuation versus frequency for the filter and

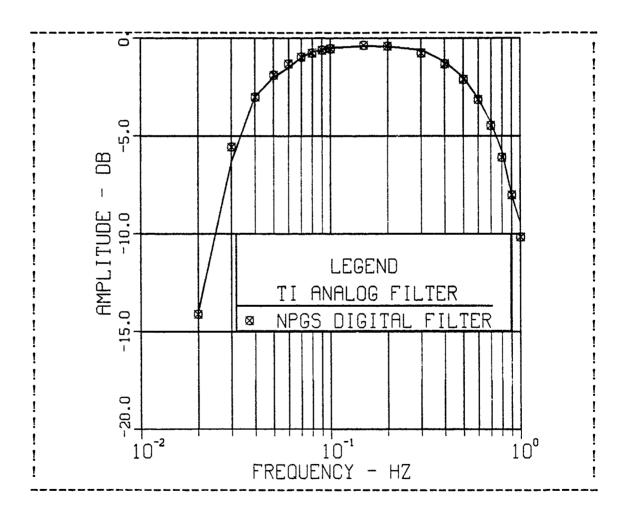


Figure 5.11: Plot of Attenuation Versus Frequency for Sinusoidal Inputs for Digital Filter and Analog Filter.

with the measured frequency performance of the AN/ASQ-81 magnetometer, which was supplied by Texas Instruments, Inc., and does not include the effects of the fixed high pass filter. Consequently, the data shown in Figure 5.11 is a comparison of the data supplied by Texas Instruments and the output of the test program, which also does not include the fixed high pass filter. As can be seen, the performance of the filter is extremely similar to that of the magnetometer itself.

3. Anderson Function Simulations

The next step in the simulation phase was introduction to the filter of Anderson function simulations. The shape of the signal amplitude of the output of magnetometer passing through the sphere of influence of magnetic anomaly (submarine) is a function of the dip angle of the geomagnetic field, the magnetic heading of the track the magnetometer (or the aircraft), the magnetic heading of the anomaly (submarine) dipole, and lateral range between the magnetometer (aircraft) and the anomaly. Anderson functions [Ref. 9] are mathematical representations of three basic components of signals which, when taken in various linear combinations, describe shape of these anomaly signals. The equations the Anderson functions are:

or, a dimensionless parameter defined as the distance traveled along the magnetome er (aircraft) track divided by the slant range at closestpoint of approach (CPA)

(Second Anderson Function)
$$f = B \times f$$

1 0 2
(Third Anderson Function) $f = B \times f = B \times f$

The Anderson functions were introduced into filter program in a noise-free signal environment in order observe the output signal and ensure that it was a "MADlike" signal. A rigorous determination of the actual output signal would have been extremely difficult to obtain, comparison was made with the output of a computer simulation Joe Rice of provided to NPS by Mr. Texas program the sampling rate of the program Instruments. When to equal that of the Texas Instruments program, 8 adjusted HZ. the two program outputs were observed to be The Anderson function simulation inputs outputs of the program are depicted in Figures 5.12 through The Texas Instruments program outputs were obtained 5.18. the form of time series plots of discontinuous and were therefore not conducive to replotting for points, comparison.

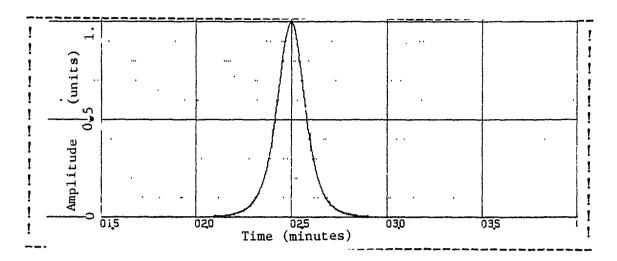
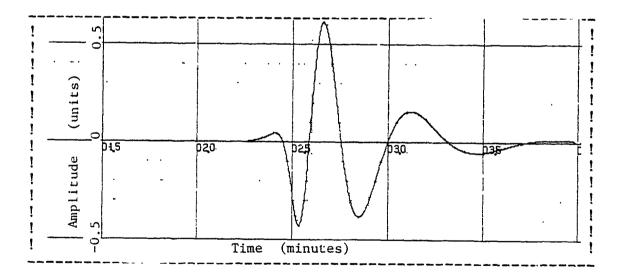


Figure 5.12: First Anderson Function Input. CPA at Time 2.5 Minutes.



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Figure 5.13: Filter Output for First Anderson Function Input.

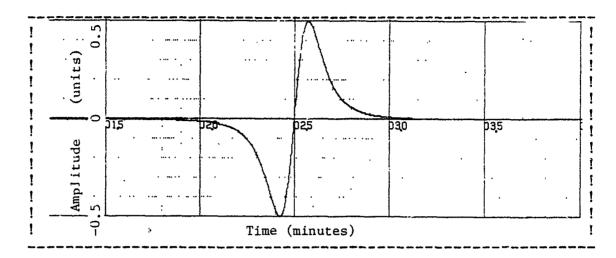
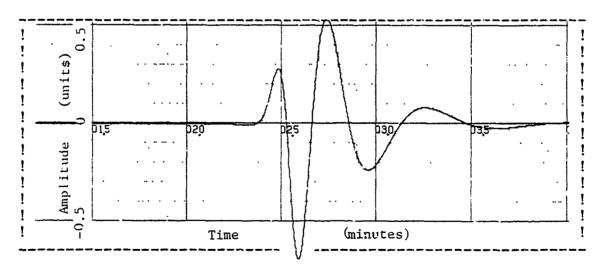


Figure 5.14: Second Anderson Function Input. CPA at Time 2.5 Minutes.



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Figure 5.15: Filter Output for Second Anderson Function.

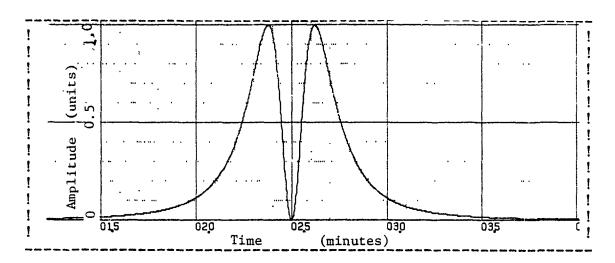


Figure 5.16: Third Anderson Function Input. CPA at Time 2.5 Minutes.

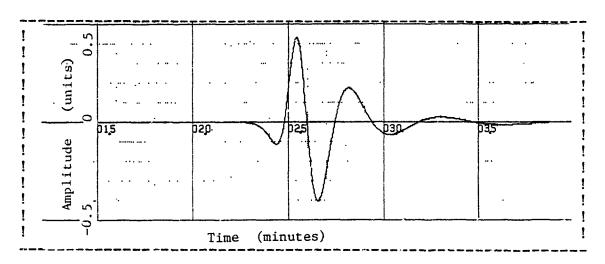


Figure 5.17: Filter Output for Third Anderson Function.

The filter output for all three Anderson function inputs did appear to be "MAD-like" signals, and did closely resemble the simulation output obtained from Texas Instruments, Inc.

4. <u>Impulse Function Response</u>

The response of the filter program was also observed when the input was a unit impulse function. Again, the

output was compared to that of the Texas Instruments' computer program. The outputs of the two programs were observed to be, again, very similar, as can be seen in Figure 5.18, where the response of the NPGS filter is represented by a solid line and that of the Texas Instruments filter by a chain-dash line. The abrupt "jumps"

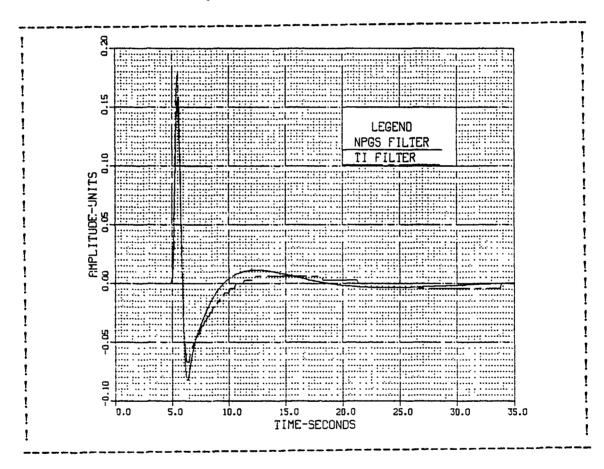


Figure 5.18: Impulse Response of Filters.

of the Texas Instruments response are due to the translation of the output plot supplied to this plot. The plot supplied by Texas Instruments was, again, discontinuous points of poor resolution, and it was necessary to interpolate values

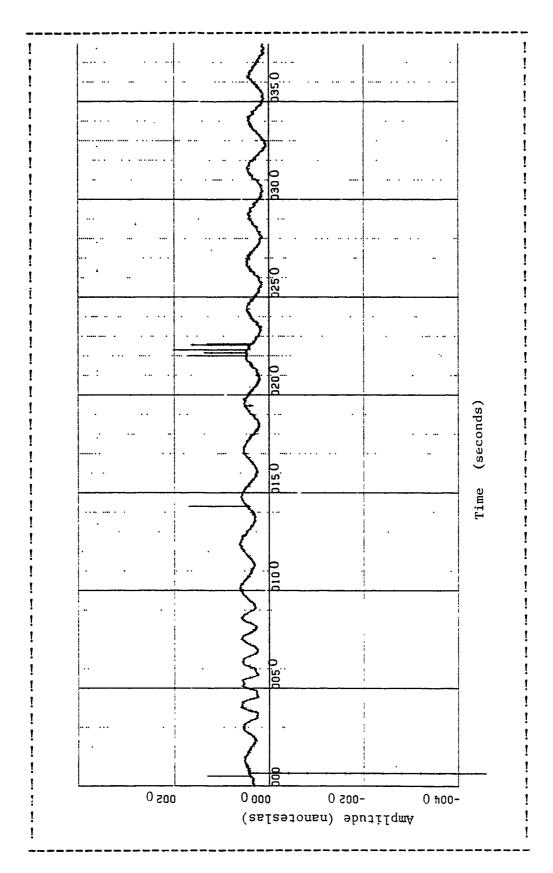
in order to generate Figure 5.18. This resulted in the broken appearance of the plot. Even so, the similarity of the outputs can be observed.

B. EQUIPMENT SETUP

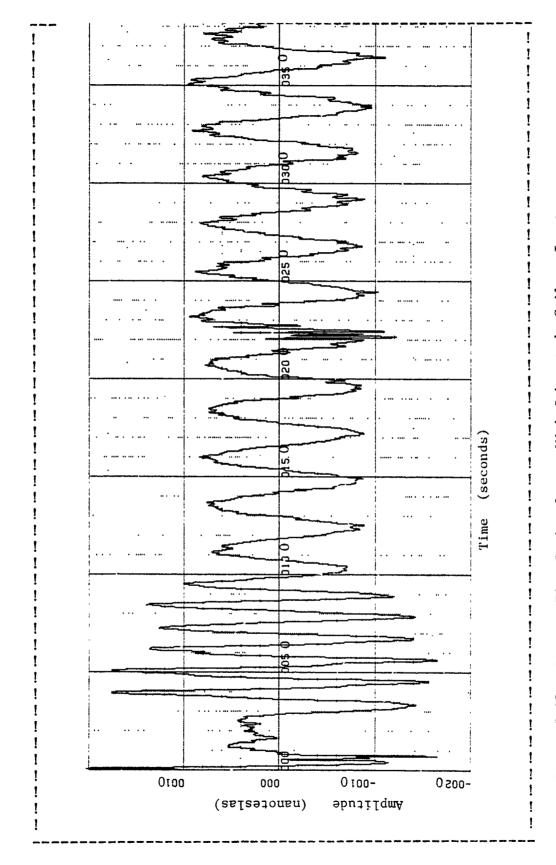
Following the simulation phase of the experiment, actual magnetic field measurements were introduced to the filter in order to test the response of the filter. Magnetic field measurements were made at the La Mesa field test site near the Naval Postgraduate School in Monterey. The output of an AN/ASQ-81 magnetometer, a Schonstedt magnetic field sensor, and the school's coil sensor, oriented along the Earth's magnetic field, were pulse code modulated (PCM) and transmitted via VHF radio to recording devices at the Postgraduate school. The recording of a two hour long data collection period was transferred to digital data tape for use by the school's IBM3033 general purpose mainframe computer.

In the first test of the digital filter program, the output of the Schonstedt sensor, which represents fluctuations of the Earth's total field, was used as the input to the computer program. A comparison of the output of the computer program, with this approximation to the total field fluctuations as input, to the output of the AN/ASQ-81 should provide an indication of the proper functioning of the computer filter program. The results of

the test are shown in Figures 5.19 through 5.21 on the pages following. Figure 5.19, the Schonstedt sensor output, shows several instances of PCM dropouts, that is, occasions where pulse code modulation signal was not correctly read the computer for some reason. At such occurances, the data point value used by the computer is a random number and does not reflect the true value of the data. The problem with these PCM dropouts is that the computer does not recognize invalid data points and will use them in them This can (and does) cause problems in computation of Fourier transforms, spectral characteristics, Additionally, this will also impact the functioning of the digital filter program which is subject of this thesis. PCM dropouts are visible at times 6, 8, 142, and 220 through 226 seconds on the plot of the Schonstedt sensor output. An examination of Figure 5.20, the filter program output, reveals the programs attempt to "follow" these PCM dropouts. It should be recalled simulations indicated the filter's tendency previous sudden changes in the input signal, relaxation time required for the filter to steady out. effect is apparent in the output of the filter program times corresponding to those of the PCM dropouts in Schonstedt sensor's time series plot. It can be seen this overshoot tendency resulted in an output significantly different from the actual AN/ASQ-81 output at these times.



gure 5.19: Schonstedt Coil Time Series Output



Program Time Series Output With Schonstedt Coil as Input Figure 5.20:

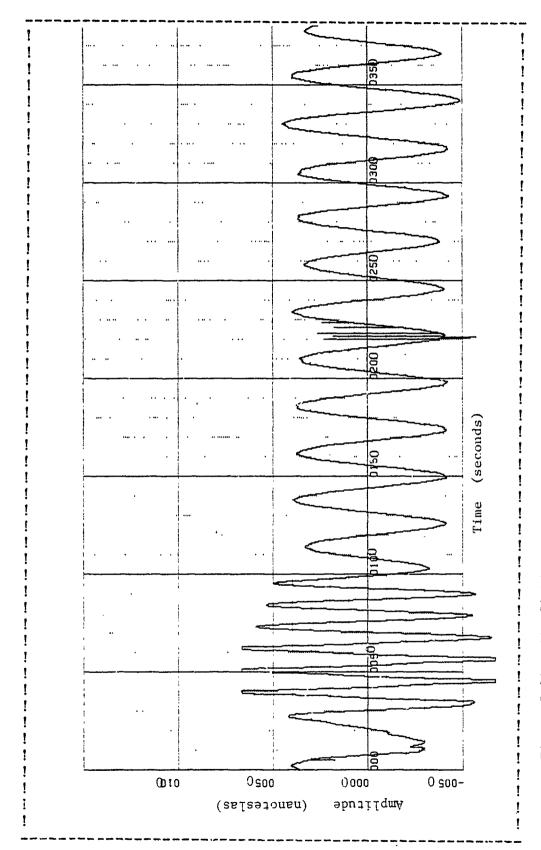


Figure 5.21: ASQ-81 Time Series Output

If the PCM dropout induced differences are neglected, it can be seen that the shape of the output of the filter program is remarkably similar to that of the AN/ASQ-81 magnetometer, although noisier. Note that the output of the AN/ASQ-81 magnetometer exceded the maximum voltage amplitude which the pre-amplifiers of the data collection system were able to handle and resulted in a truncated signal from time 40 to time 60 seconds. It can still be seen, however, that the filter program output is very similar to the time signal which would have been displayed without this truncation.

It should be noted that the amplitudes of the time series signals of the program output and the AN/ASQ-81 magnetometer differ considerably. In the case of the program output, the peak amplitudes are on the order of 1.4 nanoteslas along the vertical scale, while the peak amplitudes of the output of AN/ASQ-81 magnetometer are on the order of 0.7 nanoteslas along the vertical scale. This is because input signal to the filter program is an approximation the total field difference time series signal. amplitude difference could reasonably be expected. The intent of this initial test was to investigate the output time series shape, and an exact correlation was expected. It is worth noting that the digital filter program will perform its function on any time series signal, regardless of units. This means that a signal

operated upon either before or after conversion from whatever units it was originally measured to magnetic field strength units.

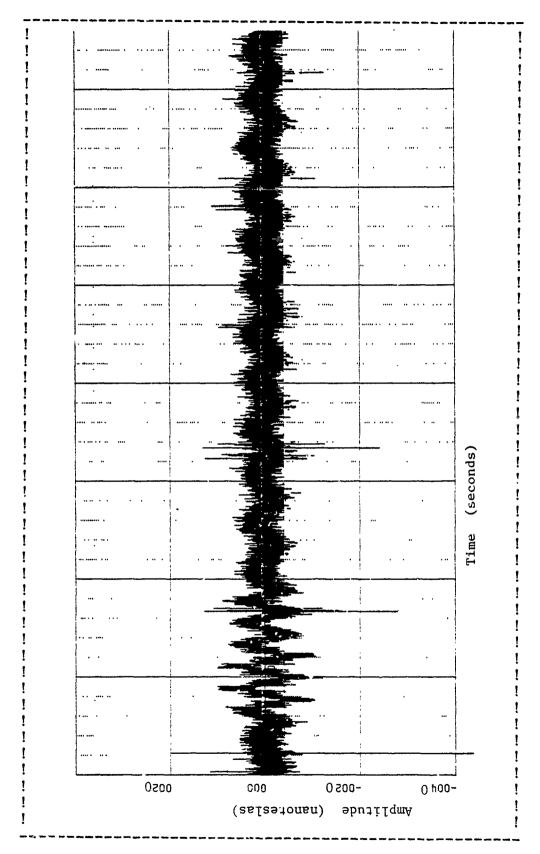
Therefore it appears that the digital filter program is functioning properly. When a close approximation to the fluctuations of the total field time series signal is used as the input to the computer program, the output of the program is similar to the time series output of an AN/ASQ-81 magnetometer.

The final stage in the testing process was a conversion of the time series output voltage signal of the coil antenna sensor, which was aligned along the Earth's magnetic field, into a total field fluctuation time series representation for the same time period as before, and then to use this as the input to the digital filter program. A comparison of the resultant time series output of the program with the actual AN/ASQ-81 magnetometer output would validate the proper functioning of the program.

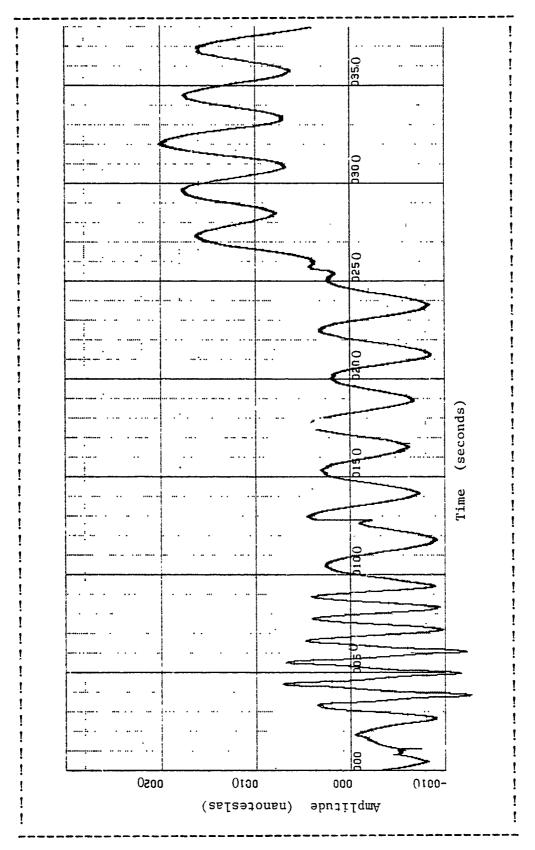
Conversion of the time series antenna sensor output voltage signal into total field fluctuations in nanoteslas was accomplished through the use of a computer program designed by Capt. Kurt Stevens, USAF, a student at the Naval Postgraduate School, as his Master's thesis [Ref 10]. The output voltage time series is stored in an array, then a Fourier transform is performed on the stored data, resulting in the Fourier spectrum of the data. This spectrum is

corrected for the characteristics of the coil antenna sensor to obtain the Fourier spectrum of the total field data. A reverse Fourier transform gives the time series signal for total magnetic field in nanoteslas.

This time series signal was used as the input the digital filter program and compared with the output of the AN/ASQ-81 magnetometer. Figures 5.22 through 5.24 show the raw coil antenna data, the total field time series data, and the program output time series for a 6 minute period of the Figure 5.22 shows the raw coil antenna data series. The number of PCM dropouts should be noted, as these will influence the performance of the filter program. 5.23 shows the computed total field time series. Note that the PCM dropouts evident on the raw time series plot evident on the computed total firly time series plot also, and thus inputted to the filte program as valid data Additionally, there are two "jumps" in the plot of points. field fluctuation (Figure 5.23) which are also total inputted to the filter program as valid data points. "jumps" are located at 128 and 256 seconds and are caused by the method of processing blocks of data for the conversion to total field fluctuation. A block of 128 seconds of data processed at a time, and the results of each block stored in an array. This results in a slight amplitude difference between the last data point of one block and the



igure 5.22: Ray Coil Antenna Time Series Output



re 5.23; Coil Antenna Difference Field Time Series Output

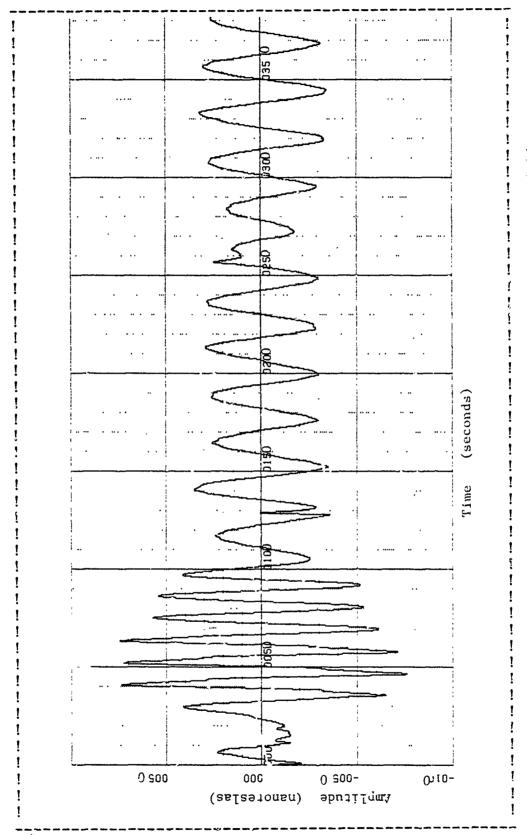


Figure 5.24: Program Time Series Output With Coil Anterna Difference Field Time Series as Input.

first data point of the next block of data. This slight difference is manifested as a signal jump.

A comparison of Figures 5.24 (program output) and 5.21 (AN/ASQ-81 output) show that the filter program gives a time series output very similar to that of the actual magnetometer. The first 20 seconds of the program output is somewhat dissimilar to that of the AN/ASQ-81, due either to the initial "start up" delay of the filter program or to distortion of the total field fluctuation time series. There is a PCM dropout at time 11 seconds which contributed to the distortion.

Following this, however, it can be seen that the program output is very similar to that of the magnetometer, except at 128 and 256 seconds, which show the effects of the false caused by the total field fluctuation jumps There is also a noticeable four to five second conversion. time delay between the AN/ASQ-81 output and that of evident filter program. As this delay is not comparison of the AN/ASQ-81 output and that of the filter program with the Schonstedt sensor as the input, it can inferred that this time delay is caused either program which converts the raw coil data to total fluctuation data, or by a phase (and hence time) change of the voltage signal due to the coil sensor itself. comparison of the raw coil data in Figure 5.22 to converted coil data in Figure 5.23 indicates no time shift,

and hence the deduction can be made that there is a time delay inherent within the coil sensor itself.

Other than the differences c the four to five second time delay and the distortions caused by the false signal jumps, the program output is extremely similar to the output of the AN/ASQ-81 magnetometer.

VI. CONCLUSIONS

The intent of this thesis was to design and test a digital filter computer program which would, when given a time series input of fluctuations in the total magnetic field, deliver an output time series representation of the output of an AN/ASQ-81 magnetometer. This purpose has been realized.

The computer program contained in Appendix I has been proven to output a time series signal which is very similar to that of the magnetometer. The major limitations of the output signal are a finite time delay of about five seconds between the AN/ASQ-81 magnetometer signal and the output signal of the program, a sensitivity of the program to false data points such as those caused by PCM dropouts and false signal jumps caused by processing large data blocks, and the inherent limitations of the program caused by its dependence on the use of digital data tapes and the IBM 3033 mainframe computer.

The five second time delay is not considered to be an important limitation to the program, as it was intended as a research tool for programs currently in progress at the Naval Postgraduate School. Instances where this time delay might become important would be in areas of simultaneous comparison of the program output signal with an actual magnetometer, in target location algorithms using time

delays, or in correlation studies between different sensors. In correlation studies using coil sensors, the effects of the time delay would cancel out, as all coil outputs would be similarly delayed. In target location algorithms, the target location errors due to the time delay could be adjusted for simply while in computer simulation, and flight testing could not reasonably be accomplished without the use of an actual magnetometer as the sensor. Lastly, in a comparison of the program output with an actual sensor, the time delay can, again, be compensated for. In short, these limitations are not considered excessive, especially as the apparent cause for the delay is not the filter program.

In the primarily intended purpose of the filter program, magnetic noise studies, the time delay is not considered to be a problem.

problem of false data points caused by PCM dropouts jumps due to conversion to total False data fluctuations is more serious. points cause inaccuracies in the output time series and could adversely Unfortunately the PCM dropout later projects. one which is endemic to the data collection system presently being used at the postgraduate school, not to the filter program itself. It is imperative users of this program are aware of the PCM dropout problem effects it may entail upon their specific research. A large number of PCM dropouts in a time series could render that series unusable. Similarly the case of the false data jumps caused by conversion to total field fluctuations is not within the filter program. Further investigation of this problem is necessary in order to eliminate it.

The last problem, that of reliance upon the digital data tape/IBM 3033 computer system, is, like the PCM problem, one which is not endimic to the filter program but rather to the data collection system being used. A change of data collection system may, at some future time, remove the reliance upon the PCM/digital tape/IBM 3033 data system (and hence too the data block conversion problem which results in falso data jumps), but this is unlikely at this time. Users should be aware of this dependence and of possible effects upon specific research projects.

APPENDIX A

AN/ASQ-81 FILTER TRANSFER FUNCTIONS

Fixed High Pass Transfer Function:

 $H(S) = \frac{1}{80 S^2 + 20 S + 1}$

Selectable High Pass Transfer Functions:

A. 0.04 HZ

H(S) =

40.82834 S

45,28317 S

40.82834 S 45.28317 S 40.82834 S + 12.52096 S + 1 45.28317 S + 11.00999 S + 1

57.576688 S

 $57.576688 \text{ S}^2 + 7.41498 \text{ S} + 1$

 $B \quad 0.06 \text{ HZ} \qquad \text{H(S)} =$

18.14591 S

20.12587 S

 $\frac{x}{18.14591}$ $\frac{x}{s^2}$ + 8.34727 S + 1 20.12587 $\frac{x}{s^2}$ + 7.33999 S + 1

25.58964 S

 $25.58964 \text{ S}^2 + 4.94332 \text{ S} + 1$

C. <u>0.08 HZ</u>

10.20708 S

11.32080 S

10.20708 S² + 6.26045 S + 1 11.32080 S² + 5.50500 S + 1

14.39417 S

 $14.39417 \text{ S}^2 + 3.70749 \text{ S} + 1$

D.
$$0.10 \text{ HZ}$$
 10.50 H 10.50 H

Selectable Low Pass Transfer Functions

A.
$$0.2 \text{ HZ}$$
 H(S) =

$$\frac{1}{0.3143 \text{ S}^2 + 1.0741 \text{ S} + 1}$$

$$\frac{1}{0.2501 \text{ S}^2 + 0.6209 \text{ S} + 1}$$

B.
$$0.4 \text{ HZ}$$
 H(S) =
$$\frac{1}{0.07858 \text{ S}^2 + 0.53706 \text{ S} + 1} \times \frac{1}{0.06252 \text{ S}^2 + 0.31044 \text{ S} + 1}$$

C.
$$0.6 \text{ HZ}$$
 H(S) =
$$\frac{1}{0.03492 \text{ S}^2 + 0.35804 \text{ S} + 1} \times \frac{1}{0.02779 \text{ S}^2 + 0.20696 \text{ S} + 1}$$

APPENDIX B

AN/ASQ-81 Z TRANSFORM FILTER TRANSFER FUNCTIONS FOR DIRECT FORM I REALIZATION

For fixed high pass filter:

$$H(Z) = \frac{-1}{1 - AFHP1*Z} + BFHP2*Z$$

$$H(Z) = \frac{-1}{1 - AFHP1*Z} - AFHP2*Z$$

where BFHPO, EFHP1, BFHP2, AFHP1, AFHP2 are constants tabulated in Appendix D.

For selectable high pass filter:

BSHPO + BSHP1*Z + BSHP2*Z + BSHP3*Z

$$- ASHP4*Z^{-4} - ASHP5*Z^{-5} - ASHP6*Z^{-6}$$

where, for low frequency cutoff of 0.04 HZ:

BSHPO = 0.99471378

BSHP1 = -5.9682827 ASHP1 = 5.9894021

BSHP2 = 14.920707 ASHP2 = -14.947051

BSHP3 = -19.894276 ASHP3 = 19.894225

BSHP4 = 14.920707 ASHP4 = -14.894327

BSHP5 = -5.9682817 ASHP5 = 5.9472141

BSHP6 = 0.99471372 ASHP6 = -0.98945296

For low frequency cutoff of 0.06 HZ:

BSHP0 = 0.9920813

BSHP1 = -5.9524928 ASHP1 = 5.9841070

BSHP2 = 14.881232 ASHP2 = -14.920650

BSHP3 = -19.841643 ASHP3 = 19.841528

BSHP4 = 14.881232 ASHP4 = -14.841757

BSHP5 = -5.9524927 ASHP5 = 5.9209919

BSHP6 = 0.99208212 ASHP6 = -0.98422128

For low frequency cutoff of 0.08 HZ:

BSHPO = 0.98945806

BSHP1 = -5.9367483 ASHP1 = 5.9788144

BSHP2 = 14.841871 ASHP2 = -14.894276

BSHP3 = -19.789161 ASHP3 = 19.788959

BSHP4 = 14.841871 ASHP4 = -14.789365

BSHP5 = -5.9367476 ASHP5 = 5.8948841

BSHP(= 0.98945802 ASHP6 = -0.97901720

For low frequency cutoff of 0.1 HZ:

BSHPO = 0.98684156

BSHP1 = -5.9210493 ASHP1 = 5.9735244

BSHP2 = 14.802623 ASHP2 = -14.867941

BSHP3 = -19.736831 ASHP3 = 19.736516

BSHP4 = 14.802623 ASHP4 = -14.737149

BSHP5 = -5.9210491 ASHP5 = 5.8688889

BSHP6 = 0.9868415 ASHP6 = -0.97384065

For selectable low pass filter: -1 -2 -3 -4 BSLPO + BSLP1*Z + BSLP2*Z + BSLP3*Z + BSLP4*Z $1 - ASLP1*Z^{-1} - ASLP2*Z^{-2} - ASLP3*Z^{-3} - ASLP4*Z^{-4}$ where BSLPO, BSLP1, BSLP2, BSLP3, BSLP4, ASLP1, ASLP2, ASLP3, ASLP4 are constants tabulated in Appendix D.

APPENDIX C

AN/ASQ-81 Z TRANSFORM FILTER TRANSFER FUNCTIONS DIRECT FORM II REALIZATION

For fixed high pass filter:

$$H(Z) = \frac{BFHPO + BFHP1*Z + BFHP2*Z}{1 - AFHP1*Z^{-1} - AFHP2*Z^{-2}}$$

where BFHPO, BFHP1, BFHP2, AFHP1, AFHP2 are constants tabulated in Appendix D.

For selectable high pass filter:

where ASHP1, ASHP2, ASHP3, ASHP4, ASHP5, ASHP6, ASHP7 are constants and tabulated in Appendix D.

For selectable low pass filter: -1 -2 -3 RCIPO + RCIP1*7 -3 RCIPO + RCIP1*7 -3

$$H(Z) = \frac{BSLP0 + BSLP1*Z + BSLP2*Z + BSLP3*Z + BSLP4*Z}{1 - ASLP1*Z^{-1} - ASLP2*Z^{-2} - ASLP3*Z^{-3} - ASLP4*Z^{-4}}$$

where BSLP0, BSLP1, BSLP2, BSLP3, BSLP4, ASLP1, ASLP2, ASLP3, ASLP4 are constants tabulated in Appendix D.

APPENDIX D

Z TRANSFORM REALIZATION DIFFERENCE EQUATIONS

With reference to Figures 3.2 and 4.2, the following difference equations are used to model the AN/ASQ-81 magnetometer filter transfer functions. The input to the fixed high pass filter is called SIG(I), where I is the current data sample. The output of the fixed high pass filter, which is the input to the selectable high pass filter, is YO(I), and the output of the selectable high pass filter, the input to the selectable low pass filter, is called YPO(I). The output of the filter is called ASQ(I). (I-1) denotes a time delay of one sample, and so forth, and the symbol * denotes multiplication.

For the fixed high pass filter:

YO(I)=BFHPO*SIG(I) + BFHP1*SIG(I-1) + BFHP2*SIG(I-2) + AFHP1*YO(I-1) + AFHP2*YO(I-2) where:

BFHPO = 0.9980499222938581

BFHP1 = -1.9960998445877161 AFHP1 = 1.9960983216843922

BFHP2 = 0.998049922238581 AFHP2 = -0.9961013674910398

For the selectable high pass filters:

XI(I) = ASHP1*YO(I) + ASHP2*XI(I-2) + ASHP3*XI(I-3)

XII(I) = XI(I) + XI(I-2) - 2*XI(I-1)

XIII(I) = XII(I) + ASHP4*XIII(I-1) + ASHP5*XIII(I-2)

XIV(I) = XIII(I) - 2*XIII(I-1) + XIII(I-2)

XV(I) = XIV(I) + ASHP6*XV(I-1) + ASHP7*XV(I-2)

YPO(I) = XV(I) - 2*XV(I-1) + XV(I-2)

For the low frequency cutoff at 0.04 HZ:

ASHP1 = 0.994713789347288 ASHP2 = -0.9952196910157882

ASHP5 = -0.9962082013015601 ASHP6 = 1.9979855321466768

ASHP7 = -0.9979897681491607

For the low frequency cuto i at 0.06 HZ:

ASHP1 = 0.9920821277199393 ASHP2 = -0.9928381306174365

ASHP5 = 1.9928247245317958 ASHP4 = 1.9943056083414792

ASHP5 = -0.9943177045263504 ASHP6 = 1.9969766455640039

ASHP7 = -0.9969861717656565

For the low frequency cutoff at 0.08 HZ:

ASHP1 = 0.9894580558875787 ASHP2 = -0.9904622611337722

ASHP3 = 1.9904384565912725 ASHP4 = 1.9924092959984783

ASHP5 = -0.9924307799269113 ASHP6 = 1.9959666590811389

ASHP7 = -0.9959835860201267

For the low frequency cutoff at 0.10 HZ:

ASHP1 = 0.9869415560096681 ASHP2 = -0.9880929619779491

ASHP3 = 1.9880549117849344 ASHP4 = 1.9905139074397893

ASHP5 = -0.9905474442556225 ASHP6 = 1.9949555824611562

ASHP7 = -0.9949820174650785

For the selectable low pass filters:

ASQ(I) = ASLP1*ASQ(I-1) + ASLP2*ASQ(I-2) + ASLP3*ASQ(I-3)

 $+ \Delta SLP4*ASQ(I-4) + BSLP0*YPO(I) + BSLP1*YFO(I-1)$

+ BSLP2*YPO(I-2) + BSLP3*YPO(I-3) + BSLP4*YPO(I-4)

For the high frequency cutoff at 0.2 KZ:

BSLPO = 0.0000000452616229

BSLP2 = 0.0000002715697375 LSLP2 = -5.7285022156249328

BSLP4 = 0.0000000452616229 ASLP4 = -0.9119356638511430

For the high frequency cutoff at 0.4 HZ:

BSLPO = 0.0000006918001209

BSLP1 = 0.0000027672004837 ASLP1 = 3.8173771378993420

BSLP2 = 0.0000041508007256 ASLP2 = -5.4670046844062743

PSLP4 = 0.0000006918001209 ASLP4 = -0.8316389680077603

For the high frequency cutoff at 0.6 HZ:

BSLPO = 0.0000033463317975

BSLP1 = 0.0000133853271900 ASLP1 = 3.7274299052305002

BSLP2 = 0.0000200779907850 ASLP2 = -5.2152772583906819

BSLP3 = 0.0000133853271900 ASLP3 = 3.2462216641520829

BSLP4 = 0.0000033463317975 ASLP4 = -0.7584278523006615

APPENDIX E

DIGITAL SOFTWARE FOR SIMULATION - DIRECT FORM (SINUSOIDS AS INPUT)

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DIGITAL SOFTWARE FOR SIMULATION - CASCADE FORM (SINUSOIDS AS INPUT)

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ONLY ONE PLOT ON THIS GRAPH, X AXIS WILL LABELLED MAGNITUDE "ON A LINEAR SCALE

ASLP63#ASQ(13)+ASLP64#ASQ(14) SLP62#YPQ(12) I-0.5) .5*SIN(2.*PI*FLOAT(I)+PHI2)+ANDISE)/600. (I)+BFHP1*SIG(I))+BFHP2*SIG(I2)+AFHP1*YG(II) SIGNAL NOISE OF +/- 0.25 TO THE OF F=10 HZ OF MAG. +/- 0. +ASHP45*XIII(12, I(12) P47*XV(12) +ASHP42 + XI(121+ASHP43 +XI(II STATEMENT ADDS TO A SINUSOID S NOISE ADDITION

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100 CONTINUE

CUTPUT PLOT COMPUTATIONS FINISHED AND ANSWERS STORED IN ARRAYS,

CALL SUBROUTINE DRAW FOR FIRST GRAPH, A TIME SERIE. REPRESENTATIONOF THE INPUT SIGNAL TO THE PROGRAM

CALL DRAW(3000, TIME, YO, 0, 0, LABEL, TITLD, 0, 0, 0, 0, 0, 5, 4, 1, LAST SCALE A LINEAR Y AXIS WILL BE TIME AND LABELLED "MINUTES" ON

END OF FIRST PLOT, PLUT SECOND PLOT

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CALL DRAW(3000, TIME, YPG, 0, LABEL, TITLE, 0, 0, 0, 0, 0, 5, 4, 1, LAST)

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APPENDIX G DIGITAL SOFTWARE FOR COMPUTATION OF SYSTEM AMPLITUDE VERSUS FREQUENCY

JSS A THE 出 18FHP1,8FHP2,4,8,C,0,E,F 14H, 11,21,K11, A1,881,221,DD1,EE1,FF1,GG1,HH1,1 PROGRAM ANGE OF 30001,TIME(3000) 201 30001,XIV(30001,XV(3000 SHP46, ASHP MGN4 FORTRAN
TER PROGRAM FOR THE ASQ-81 AND OBTAIN THE ITC FOR COMPARISON WITH MEASURED DB LOSSE ETUMETER. A SINGLE FREQUENCY SIGNAL WILL OUTPUT DIVIDED BY THE RMS INPUT TO DETER Ľ EQUENCY R 20. ⋖ **2/320.1 /8.+T**2/ WITHIN THE FR , ASHP45 , ESL P FILTE PASS TRU(3 RED(2 III(3 SLP64 SLP64 SLP63 SMSQT FFICIENT 11+ SS 000 X X PA: +1/8 8-1-E HI AAB BAA SUMSQ. SIGNAL M HI GH SQ(30 00(300) 1 (300) BEHPO FF A SHP4 COE できる。 そんもので でんない OL. ELECTABL T**2/1600-201/(
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//10+7/80+7**2/ AFH P21 BFH L11 SHP 42. ASH SSLP 62. ASL 3SLP 61. BSL D0 D SEED. S THIS PROGRAM IS DESIGNED IGITAL FILTER PROGRAM CHARACTERISTIC FOR COMFASO BI MAGNETUMETER. AND THE RMS OUTPUT DIVISET UP ARRAYS. SIG! COND FIXED 30001 . YPC 30001 . YPC 30001 . XI COMPUTE PER SEC SIL 8283 831 ,1106) FOR DI MENSION DI MENSION DI MENSION PEAL*8 REAL*8 TO SEED REAL*8 AND BESCON AND BESCON RECON AND BESCON RECON AND BESCON AND BES FOR 15.45 15.45 15.45 Nm. 2096/ 00988 00999/ 5.2831 EXEC FRTXCLGP DRT.SYSIN DD * ENT CAS FINE AND じころう F 1 3 1 1 1 SCI COEFFICE 0 出土 ら404 AFHP1: BFHP0: BFHP0: BFHP1: ` 3 7 નંનનન ш TEN 4800 000 37

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APPENDIX H

DIGITAL SOFTWARE FOR SIMULATION (ANDERSON FUNCTIONS AS INPUT)

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KK 1=(-2.+FF*(T**2)/2.)

LL 1=(1.0-FF*(T**2)/2.)

ASLP61:-(GG 1*KK 1+HH1**A2)/4.)

ASLP62:-(HH1**L1+1H1**K1+111*JJ1)/ASLP63:-(HH1**L1+111*K1+111*JJ1)/ASLP63:-(HH1**L1+111*K1+111*JJ1)/ASLP63:-(AA1*DD1)/(GG1*JJ1)

BSLP61:(AA1*DD1)/(GG1*JJ1)

BSLP63:(AA1*FF1+BB1*DD1)/(GG1*JJ1)

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THIS SIGNAL IS THE FIRST ANDERSON FUNCTION. TO JBTAIN THE SECOND ANDERSON FUNCTION, MULTIPLY BY BETA, AND AGAIN TO OBTAIN THE THIRD ANDERSON FUNCTION - *XV(11) LP62*ASQ(12)+ASLP63*ASQ(13)+ASLP64*ASQ(14) P61*YPO(11)+BSLP62*YPO(12) 4*YPO(14) G(12)+AFHP1*Y0(I1 [I] +ASHP45*X II I (I2) (I I I (I2) (SHP47*XV(I 2) I(I2)+ASHP43*XI(II SIG(I) = NORM*(10, BETP) **5)

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XI (1) = ASHP41*YO(1) +ASHP42*XI(12) +ASHP43*XI(1)

XI (1) = XI (1) +X (12) -2 *XI (11) +ASHP45*XI(1)

XI (1) = XI (1) +ASHP44*XI (11) +ASHP45*XI (12)

XI (1) = XI (1) +ASHP46*XI (11) +ASHP45*XI (12)

XI (1) = XI (1) +ASHP46*XI (11) +ASHP47*XV (12)

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CAIL DRAW(2400, TIME, YD, 0, 0, LABEL, TITLD, 0, 0, 0, 0, 0, 10, 4,1, LAST) A LINEAR WILL BE TIME AND LABELLED "MINUTES" ON AXIS

END OF FIRST PLOT, PLOT SECOND PLOT SECOND PLOTWILL BE A TIME SERIES REPRESENTATION OF THE OUTPUT AS COMPUTED

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LABELLED HAVE X AXIS LABELLED "MAGGITUDE", Y AXIS BOTH ON A LINEAR SCALE ٦= • • PLOT WILL

CALL DRAW(2400, TIME, ASQ, 0, 0, LABEL, TITLB, 0, 0, 0, 0, 0, 0, 10, 4, 1, LAST) CALL DRAW(2400; TIME; YPB:0:0:LABEL; TITLE;0:0:0:0:0:0:10:4:1; LAST OF THE *TRUE * SIGNAL VERSUS TIME AND TIME ON THE Y AXIS A PLOT X AXIS THIRD PLOT WILL BE WITH SIGNAL ON THE

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APPENDIX I

DIGITAL FILTERING SOFTWARE

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| | THIS PROCEDURE FURNISHED BY DR. TIM STANTON, DEPARTMENT OF OCEANOGRAPHY. | |
|-----|---|--|
| | READ DATA FROM IUN, ALLIGN , CHECK & RETJRN | |
| | IUN=TAPE NUMBER, EG 20 IG=INTEGER#2 ARRAY, 16 LONG, (VALUES 0-4095, SUBTRACT 2048)#5 IRS= NUMBER GF RESINCS ALLOWED (ERRORS) IREC= COUNTER OF RECORDS (FRAMES OF DATA) BLOCK 512 BITS, 32 BITS = RECORD IRO= NUMBER OF ACTUAL RESINCS (ERRORS) | |
| | INTEGER * 2 IO(16), IP(16) DATA IRR / 0/ IF (IREC.EQ.0) IS=0 | |
| 0 | EK=0 FORMAT (16A2) F (1S.NE.0) GG TG 50 READ (1UN, 20, END=900) IP REC= REC+1 | |
| 0 | IS=IS+1 IF (IS-LT-17) GO TO 50 READ (IUN, 20, END-900) IP | |
| 50 | ÎREĈ=IREC+1 ICH=IMASK(1P(1S),3,0)+1 WRITE (6,55) ICH, IS, IUN, IREC FORMAT (* RESYNCING ICH, IS, IUN, IREC | |
| | IF (ICH.NE.1) GO TO 40 DO 100 I=1 16 IO(I) = I SHI FT (IP(IS) 4) ICH=IMASK (IP(IS) 3 0) +1 IF (ICH.EQ.1) GO TO 80 IER=IER+1 | |
| 0 8 | WRITE (6,70) IUN, IREC II, ICH, IER FORMAT (* UNIT *, 13, * RECORD *, 16, *CHAN & DATA CH *, 214, \$ 'ERRORS *, 17) IS=IS+1 IF (IS*LT*, 17) GO TO 100 READ (IUN, 20, END= 900) IP | |
| 100 | I REC = I REC + 1 CONT INUE | |

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THE DATA TO TRANSFORM. ů, IN PAR-INVERSE OTINES. ONLY ST ANSFORM ELE-S USED ON A TRANSFORM ARR WI H DE THE SNS ō OF œ ONTAI E œ osi⊢ •3æ S SIL 20 AND F THE CONTENTS I ONED ARRAY JF IS GIVEN BY PART அய்வ 9 INVERS W & UL A FORWARD ALSO BE U FOURIER T FORWARD A -DIMENSIONAL ARKAY CONTAINING ON OUTPUT DATA CONTAINS THE GRDER ING IS EXPECTED. THE FIFTSTEST. ں DATA ۲, **DIMENSIONS** 11 IMAGINARY 0.0. ⋖ DAT OR ш -THE N 2 4 , WORK) FORWARD `≻∢ S UPON WHETHER , 160. FOURT MAY IN WHICH CASE A NSION IN TURN. AX RANSFORMS AREORMS INITIONS FESED IN FORM ARD A REAL LE. ARR THE HO HO IGN, IFORM WHETHE • ELY THE COMPUTES THE FORM SFORM OATA A SINGLY-DINGLY HER AINING GER INDICATING WHETHER O BE PERFORMED. I GN=-1 FOR FORWARD TRI I GN=1 FOR INVERSE TRAF I THESE DEFINITIONS TICULAR, THE DEFINITRANSFORM ARE REVER ALL SET MENSIONS OF ARRAY NN. ER INDICATING V DIF DATA IS PU OTHERWISE IS SET TO OF V DATA MUST BE DEPENDS PERFORME NRRAY IN CONT S A,NN,NDIM,I ARGUMENT LEX*8 MULTI-RANSFORMED. ALFORTRANGING THE FAS ARRAY SUBROUTINE FOURT CONCOCCOLEY-TUKEY FAST FE ARRAY DATA. FOR DATA L. THE JTH COMPONENT SUM (DATA (K) * W** (M) WHERE THE SUM IS TAY WHERE THE SUM IS TAY _ 4<u>0</u> TS TO BE AENSIONAL A SLONG EAC 当出 IN mmo-**ER#4** SEQUENC AN INTEGE PURELY RE IFORM=0 IFORM=1 IFORM=1 MENTS IN DAT P. œ COMPLE BE TR NORMAL CHANGI A. SOUR! шv PTION NUMBE MENTS DURT JOON JAKAN JAKAN NOTION NATA ZS 2 THE VAL TRANSFO MULTI-E PERFORM S URPOS **15 I GN** ND I'M 1FOR Ø

THE ŧ THIS IS THE FASTEST AND MOST VERSATILE VERSION OF THE FFT KNOWN TO THE AUTHOR. A PROGRAM CALLED FOURZ IS AVAILABLE THAT ALSO SIC FORTRANS THE FAST FOURIER TRANSFORM AND IS WRITTEN IN USASI BA-SIC FORTRAN OF THE INPUT ARRAY (WHICH MUST BE COMPLEX) THE POWERS OF TWO. ANOTHER PROGRAM. CALLED FOURING ONE TENTH AS ARRAY WHOSE LENGTH IS A POWER OF TWO. ID THE TRANSFORM VALUES MUST REPRESENT HEIR RESPECTIVE DOMAINS OF TIME AND LESE SPACINGS DELTAT AND DELTAF, IT AF=2*PI/(NN/I)*DELTAT)* OF COURSE, SAME FOR EVERY DIMENSION* THE INPUT DATA AND THE TRANSFORM OUTVELES OF PERIODIC FUNCTIONS. S FOR MOST APPLICATIONS FOURT, IF COMPILED UNDER FORTRAN H, IS COMPARABLE IN SPEED AND ACCURACY TO THE IMSL FFT SUBROUTINES. WITH CERTAIN PATHOLOGICALLY ILL—CONDITIONED DATA THE ACCURACY OF FOURT MAY BE SERIOUSLY DEGRADED, BUT THE SAME CAN PROBABLY BE SAID OF ANY EXTANT FFT ROUTINE. HAN THAT REQUIRED BY THE IMSL ROUTINES OF FOURT MAY BE GREATER OR LESS THAN THAT REQUIRED BY THE IMSL ROUTINES. FOURT ALONE PROVIDES THE CAPABILITY OF TRANSFORMING A MULTI—DIMENSIONAL ARRAY WITH A SINGLE CALL. VALUE ESS THAN TRANSFORM S PRO IHI NN Q. ERROR HALTS NOIM OR ANY THE AND DR IF HE NUMBER OF INPUT DATA
MUST BE THE SAME.
BOTH THE INPUT DATA AND
EQUISPACED POINTS IN THE
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                                                                    PROGRAM BY NORMAN BRENNER FROM THE BASIC PROGRAM BY CHARLE:
RADER, JUNE 1967. THE IDEA FOR THE DIGIT REVERSAL MAS
SUGGESTED BY RALPH ALTER.
DOCUMENTATION REVISED BY JOANNE BOGART, AUGUST 1979, NPS.
REAL
                                                                                                                                                                                                                          ARE ITS FACTORS
                                                                                                     SUBROUTINE FOUR T(DATA, NN, NDIM, ISIGN, IFDRM, WORK)
DI MENSION DATA(1), NN(1), IF ACT(32), WORK(1)
DATA TWOPI/6.2831853071796/, RTHLF/0.70710678118655/
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NT OT=2
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IF (NN(IDIM))920, 2
NT OT=NT OT*NN(IDIM)
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EXAMPLE 2. ONE-DIMENSIGNAL LENGTH 64 IN FORTRAN II DI MENSIGN DATA(2,64) DATA(1,64) DATA(1,1)=REAL PART DATA(2,1)=0.
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N= NN(IDIM)
NP 2=NP 1*N
IF (N-1) 920, 900, 5
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1. COMPLEX TRANSFORM OR REAL TRANSFORM FOR THE 4TH DIMENSIONS.

2. REAL TRANSFORM FOR THE 2 ND OR 3RD DIMENSION. M TRANSFORM FOR THE 1ST DIMENSION. N ODD. M SET THE 1 THE 1 ST DIMENSION. N ODD. M SEAL TRANSFORM FOR THE 1 ST DIMENSION. N EVEN. TRANSFORM A COMPLEX ARRAY OF LENGTH N/2 WHOSE IMARETE SECOND HALF BY CONJUGATE SYMMETRY.
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                                                       DATA BY BIT REVERSAL, SINCE N=2**K. AS THE DONE BY SIMPLE INTERCHANGE, NO WORKING ARRAY
                                                                                                                                                                                                                                                                                                                                                                            ARZAY
                                                                                                                                                                                                                                                                                                                                                                            GENERAL
                                                                                                                                                                                                                                                                                                                                                                            REVERSAL
                                                                                         IF (NT WO-NP2 8200,110,110

NP2 HF = NP2/2

J= 1

DD 150 12=1; NP2,NP 1

IF (J-12)120,130,130

ILS 12+NP1-2

DD 125 13=11*NP2*NP 1

J= 12+NP1-2

DD 125 13=11*NP2*NP 1

TEMPR = DATA(13)

DATA(13)=DATA(13)

DATA(13)=DATA(13)

DATA(13)=DATA(13)

DATA(13)=TEMPR

DATA(13+1)=TEMP I

M=NP2 H 150,150,145

IF (M-N21)150,140,140

J= J+M

GO TO 300
                                                                                                                                                                                                                                                                                                                                                                                                  NW ORK = 2*N

DO 270 II=1 *NP1.2

DO 270 I3=11.NT OT. NP2

J= 13

DO 260 I=1.NWO RK, 2

IF (ICASE-3) 210, 220, 210

WORK (I+1)=DATA(J)

WORK (I+1)=DATA(J)

WORK (I+1)=0 ATA(J)

WORK (I+1)=0.

IF P2=NP2

IF P1= IFP2/IFACT(IF)

J= J+IFP1
                                                                                                                                                                                                                                                                                                                                                                             DIGIT
                                                                                                                                                                                                                                                                                                                                                                             βY
I=1
DD 80 J=1,NTDT
DATA(J)=DATA(I)
I=I+2
                                                                                                                                                                                                                                                                                                                                                                             SAUFFLE DATA
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CAN BE D
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COCOCO

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DD 630 J3=1 1,NT OT, IFP 2

TEMPR = DATA [ J31 | J31 |
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3RD

OR

REAL TRANSFORM SYMMETRIES.

ETE A

COMPLE

(II RNG-NP 1)805,900,900

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A(11+DATA(
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     1 MAX 1 750, 760, 7

1 = DATA(1 M IN) -

1 + 1 |= 0 - 780, 780
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755
71=0ATA(I)
7+1)=-6ATA
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805 DD 860 13=1.NPD 1.NPD 1.NP
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DD UNIT=3400-4,VOL=SER=MIKE1,DISP=(OLD,KEEP), LABEL=(1,NL,IN), DCB=(RECFM=FB,LRECL=32,BLKSIZE=512,DEN=2) DD SYSUUT=A,OUTLIM=65000 IN VOLTS SER IES RAW CDIL TIME S
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//GO.SYSDUMP DD

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